



JSC Commercial Human Space Flight Symposium

October 13-14, 2010

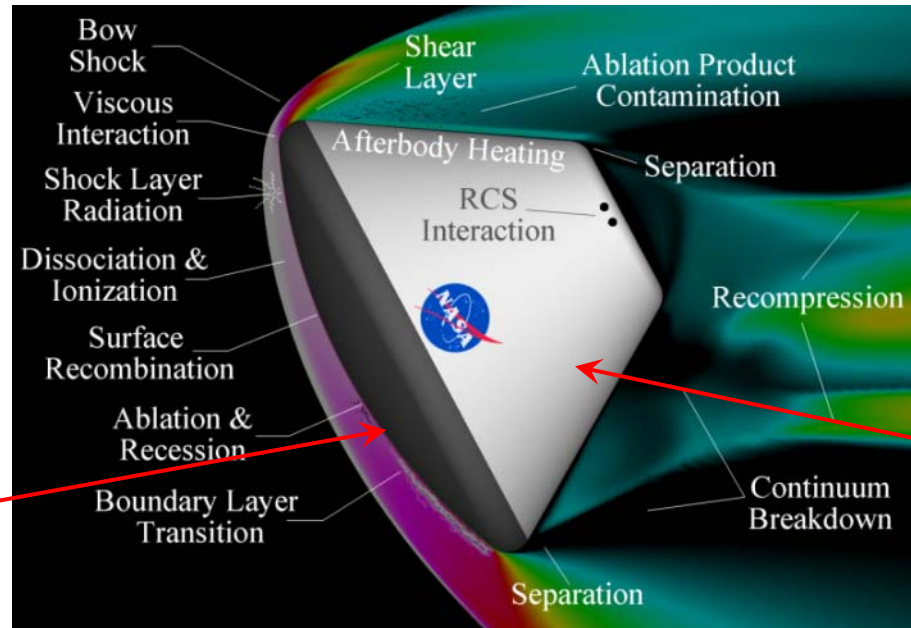
**Thermal Protection System (Heat shield) Development –
Advanced Development Project
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Orion Thermal Protection System Manager**

The Orion TPS Objective



***Enable the CEV
Project Office and
the Prime to develop
a CEV heat shield...***

Heat shield



Back shell

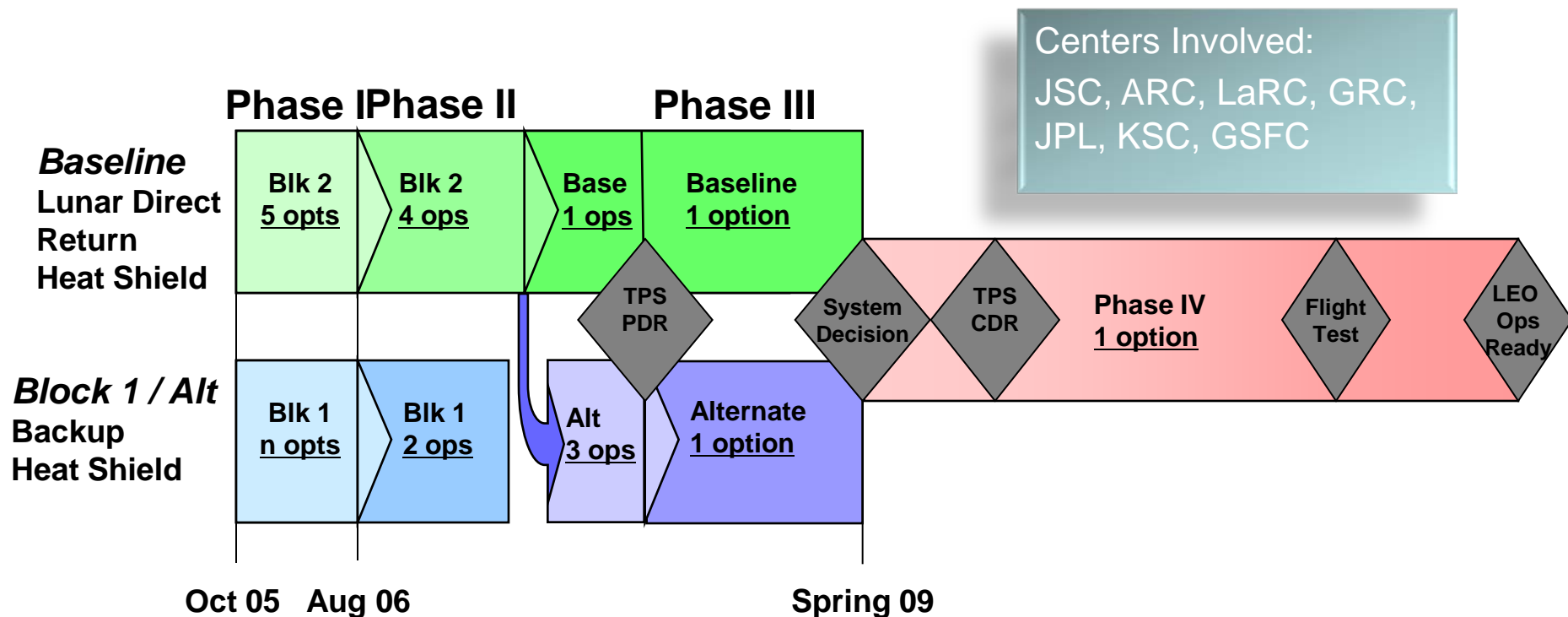
- A top risk identified at project initiation was the development of a heat shield capable of both LEO and Lunar return
 - Ablative TPS materials are required for LEO and Lunar missions
 - The US had focused little attention on ablative materials since Apollo era
 - Applicable ablative TPS materials were at low technology readiness levels (TRL ~ 3-4)
- In Oct 2005, the CEV Project commissioned the CEV TPS Advanced Development Project to address the heat shield development risk

... by initiating a Multi-Center Advanced Development Project to raise the TRL and reduce the risk of Lunar return capable ablative TPS materials and heat shield systems

CEV TPS Development Strategy (Critical Path Item)



- **Baseline** Heat shield (Lunar and LEO return capable) by Orion IOC
- **Alternate** Heat shield (Lunar and LEO return capable) parallel development, maintained up through system decision
- NASA develops **Baseline** & **Alternate** heat shield designs up to Orion PDR
- Prime takes over responsibility of heat shields after CEV PDR – w/ NASA oversight
- Back shell TPS development controlled by Orion Prime – w/ NASA oversight



Scope of TPS ADP Primary Objectives



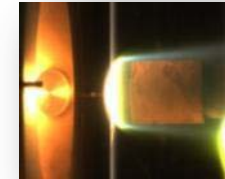
- **TPS materials fabrication and characterization**

- Development of material constituent, processing and properties specifications
- Detailed mechanical and thermal material properties testing



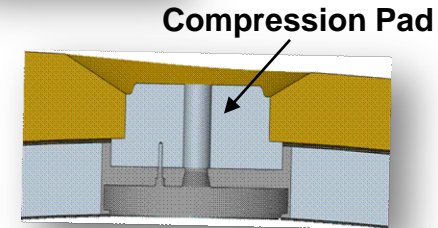
- **TPS materials thermal performance capabilities for LEO & Lunar returns**

- Screening and comprehensive TPS materials thermal performance testing
- TPS materials thermal response models
- TPS thermal performance margins policy



- **TPS materials thermal-mechanical performance capabilities**

- Ground, launch, on-orbit, nominal and emergency entry, descent & landing loads
- Thermal-structural integrated (carrier structure + TPS) testing



- **Design for all heat shield components**

- TPS acreage, carrier-structure, TPS bonding, compression pads, main seals, gap/seams, close-outs, repairs

- **Integrated heat shield design and performance capabilities**

- Integrated FEM analysis and design of all components
- TPS material thermal, MMOD and integrated sizing
- Integrated thermal-structural analysis and design of complete heat shield

- **Manufacturing for an integrated 5 meter heat shield**

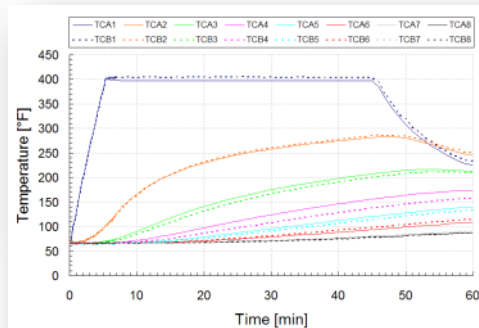
- Infrastructure, staffing, resources and equipment for full-scale heat shield production
- Demonstration of full-scale heat shield manufacturing procedures



Heat shield Thermal-Structural Test and Analysis



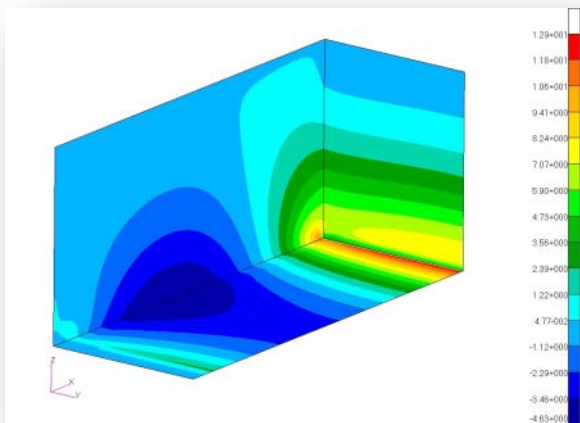
4-point Bend Flexure Test



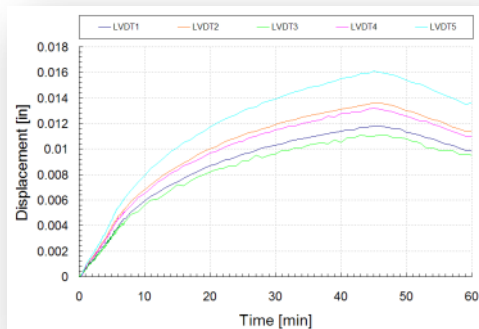
Temperature Transients



**Thermal Vacuum Cycling-
Simultaneous thermal & vacuum exposure**



Pre-test Analysis Predictions



Displacement Measurements

Testing performed
primarily at LaRC



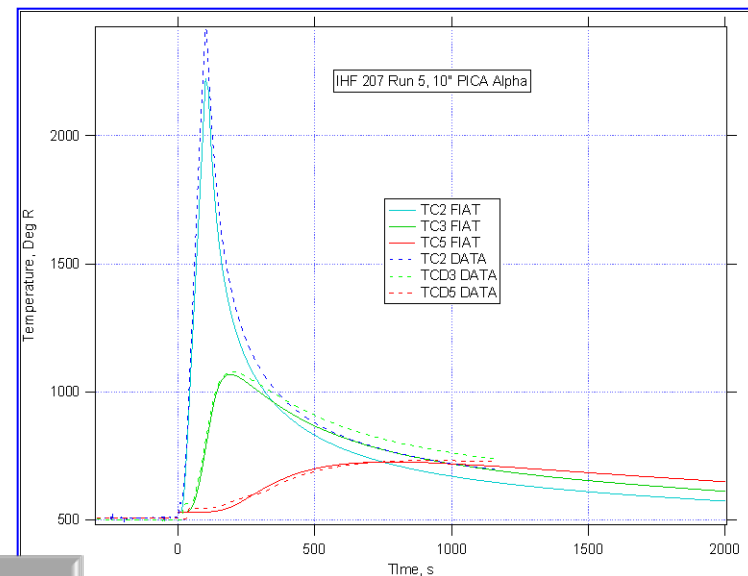
**Environmental Chamber Test
(-150 F to +250 F)**

Heat shield Arcjet Testing



Pre-test 100% N2 10% O2 23% O2 30% O2

Oxygen Sensitivity Testing



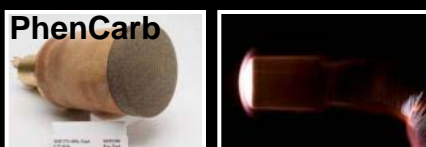
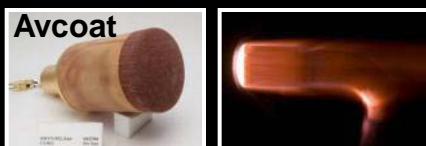
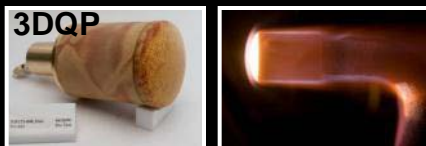
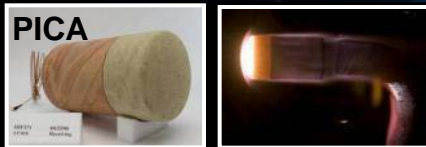
Model Correlation



Gap & Seam Testing

Testing performed at
both ARC & JSC

Direct Results of the Orion TPS ADP



Competitive materials R&D resulted in multiple viable materials & systems



Avcoat: Selected for the Orion



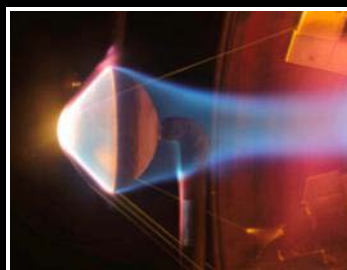
PICA: Selected for MSL & Dragon



TPS ADP Arcjet tests revealed catastrophic failure mode of initial MSL TPS



MSL shifts to a new TPS ADP developed TPS material



Large article arcjet testing demonstrated during TPS ADP is now a necessary TPS tool



- New NASA TPS experts
- Multiple TPS firms
- Large scale manufacturing
- TRL = 5-6 ablative TPS
- Promising new TPS concepts
- Technology transfer to commercial space industry

TPS ADP Completion and Transition to Prime



- The ADP matured two heat shield designs, Avcoat and PICA, to PDR status
- The final down-select between PICA & Avcoat proceeded as planned (3/31/2009)
- The TPS ADP terminated as planned on 3/31/09, and responsibility for the heat shield was transitioned to LM
- NASA continues to play a vital role in the Orion TPS design activities
 - NASA has substantial in-house capability in TPS, including expertise developed during Shuttle Return-to-Flight activities and the CEV TPS ADP, which is being leveraged for the CEV TPS development, design and certification
 - The Prime Contractor retains primary responsibility for the development, design, test, certification and delivery of the TPS
 - There are several critical areas where NASA has in-line responsibilities, including:

Analytical

- Ablation thermal response model development
- MMOD-impacted TPS response models
- Margin Policy development and buy-down
- Compression pad analysis & design

Testing

- Arcjet and radiant heat
- Thermal barrier
- Material property
- Tuneable beam shock

Conclusions from the Orion TPS ADP



- **The Orion Thermal Protection System (TPS) ADP was a 3 1/2 year effort to develop ablative TPS materials for the Orion crew capsule**
 - The ADP was motivated by the lack of available ablative TPS's
 - The TPS ADP pursued a competitive phased development strategy with succeeding rounds of development, testing and down selections
 - The Project raised the technology readiness level (TRL) of 8 different TPS materials from 5 different commercial vendors, eventual down selecting to a single material system for the Orion heat shield
- **In addition to providing a heat shield material and design for Orion on time and on budget, the Project accomplished the following:**
 - Re-invigorated TPS industry & re-established a NASA competency to respond to future TPS needs
 - Identified a potentially catastrophic problem with the planned MSL heat shield, and provided a viable, high TRL alternate heat shield design option
 - Transferred mature heat shield material and design options to the commercial space industry, including TPS technology information for the SpaceX Dragon capsule



Capability: The development of Thermal Protection Systems (TPS), including thermal & structural analysis, design, materials & processes, and thermal (arcjet and radiant heat) testing

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